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FROM: PROI (TI) (STINFO)

19 Apr 2000

SUBJECT: Authorization for Release of Technical Information, Control Number: **AFRL-PR-ED-TP-2000-080**
Liu, C.T., Yang, J.N., "Determination of Equivalent Initial Flaw Size in a Particulate Composite Material"

8th ASCE Conference on Probabilistic Mechanics and Structural Reliability (Statement A)
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Determination of Equivalent Initial Flaw Size in a Particulate Composite Material

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Abstract

In this study, a method for determining the initial flaw size in a particulate composite material is developed. The results of analyses indicate that the initial flaw length and the critical flaw size follow the second asymptotic distribution of maximum value. It also indicates that on the first approximation, it can be assumed that the initial flaw size and the critical flaw size is independent of specimen thickness.

Introduction

An important engineering problem in structural design is evaluating structural integrity and reliability. It is well known that structural strength may be degraded during its design life due to mechanical or chemical aging, or a combination of these two aging mechanisms. Depending on the structural design, material type, service loading, and environmental condition, the cause and degree of strength degradation due to the different aging mechanisms differs. One of the common causes of strength degradation is the result of crack development in the structure. When cracks occur, the effects of crack sizes and the rate of growth on the fracture resistance of the material need to be investigated.

Crack propagation in particulate composite materials has been an important subject of research for many years. It has been shown in recent literature that the crack propagation rate in particulate composites is a power function of the Mode I stress intensity factor (Liu, 1990; Liu 1990; Liu and Yang, 1994). In order to predict the crack growth rate, it is necessary to determine the initial crack size.

It is well known in the aerospace industry that the initial flaw size in metals and superalloys is too small to be detected by any NDI techniques. Consequently, the initial flaw size in metals has been determined using experimental results, such as fractographic data or S-N data (Yang et. al, 1995 and Yang et. al, 1997). From the experimental S-N data, one can determine the terminal crack size (critical crack size) at the time of failure. The initial flaw size is then computed from the terminal crack size by conducting the crack growth analysis backwards.

To date, the concept of predicting the crack growth rate based on the initial flaw size and the determination of the initial flaw size in highly filled polymeric materials has not been studied yet. Therefore, it is the objective of this study to investigate the concept of predicting the crack growth rate based on the initial flaw size and to determine the initial flaw size existing in a particulate composite material.

In this study, the equivalent initial flaw size (EIFS) in a particulate composite material, containing hard particles embedded in a rubber matrix, was determined using constant strain rate crack propagation test data. Uniaxial tensile specimens with and without pre-crack were tested at a constant strain rate of 0.067 in/in/min. Four different specimen thicknesses (0.2 in., 0.5 in., 1.0 in., and 1.5 in.) were considered. The effect of specimen thickness on the EIFS was investigated and the results are discussed.

Analytical Analysis

To determine the EIFS, the following information is needed: (1) crack growth rate parameters, (2) critical stress intensity K_{IC} and threshold stress intensity factor K_{th} under which crack will not grow, and (3) time to failure data under constant strain rate. Crack growth rate parameters as well as K_{IC} and K_{th} are determined experimentally using pre-flawed specimens. Time to failure data are also obtained experimentally using specimens without a pre-flaw.

For pre-flawed specimens, the stress intensity factor K_I is given by

$$K_I = \sigma (\pi a)^{1/2} f(a/w) \quad (1)$$

In which σ is the applied stress, $f(a/w)$ is the geometric correction factor, a is the crack length, and w is the width of the specimen. The functional relationship between $f(a/w)$ and a/w is shown below.

$$f(a/w) = 0.5854(a/w)^3 + 1.099(a/w)^2 + 0.8672(a/w) + 1.049 \quad (2)$$

For a specimen subject to a constant strain rate, the stress intensity factor K_I reaches the critical stress intensity factor K_{IC} at the instant of fracture, and the corresponding flaw size is denoted by a_c , referred to as the critical flaw size or the terminal flaw size. It follows from Eq. (1) that

$$K_{IC} = \sigma_c (\pi a_c)^{1/2} f(a_c/w) \quad (3)$$

Where σ_c is the critical stress at fracture.

The crack growth rate da/dt has been shown to be a power function of the stress intensity factor K_I , i.e.,

$$da/dt = Q K_I^m \quad (4)$$

in which m and Q are crack growth rate parameters.

When a specimen without pre-flaw is subjected to a constant rate, the entire loading history and hence the stress history $\sigma = \sigma(t)$ can be measured, including the critical stress σ_c at the time of fracture, t_c . For a given critical stress intensity factor K_{IC} (material constant), the critical flaw size a_c can be computed from Eq. (3). Consequently, the initial flaw size a_0 at $t=0$ can be obtained by integrating Eq. (4), based on the terminal condition (a_c, t_c) and the stress history $\sigma(t)$.

Experimental Analysis

Constant strain rate tests were conducted on specimens with and without pre-flaws at a strain rate of 0.067 in/in/min. The critical stress σ_c and the time to failure t_c were determined from the specimen without pre-flaw. The crack growth parameters m and Q were determined from the specimens with pre-flaw. The results are: $m = 2.084$ and $Q = 9.3325 \times 10^{-7}$ in which the units are force in pound, length in inch, and time in minute. Further, the critical stress intensity factor and the threshold stress intensity factor are 78.3 psi (in)^{1/2} and 52 psi (in)^{1/2}, respectively.

Statistical Distribution of Equivalent Initial Flaw Size and Critical Flaw Size

The results of the analysis show that the equivalent initial flaw size a_0 as well as the critical flaw size a_c vary from specimen to specimen. Hence, the statistical distribution of these quantities should be determined. In this study, four statistical distribution functions, (1) normal distribution, (2) two parameter Lognormal distribution, (3) two parameter Weibull distribution and (4) second asymptotic distribution of maximum values, were considered. The goodness of fit for different distributions has been conducted using the Kolomogorov-Smírov test. The results indicate that the second asymptotic distribution of the maximum value has the best fit for the distribution of a_0 and a_c .

Results

Based on the constant strain rate test results for specimens with and without preflaws, a method for determining the equivalent initial flaw size existing in the particulate composite material has been presented. The method is based on a backward crack growth analysis starting from the given terminal flaw size to determine the equivalent initial flaw size.

Equivalent initial flaw sizes were computed for the cases in which the effect of K_{th} is ignored. Typical plots of statistical distributions of a_0 are shown in Figs.1-4. For a comparison purpose, experimental data, shown as circles, are also included in these figures. It is seen that the second asymptotic distribution of maximum value fits the experimental data the best that is consistent with the results of the goodness of fit analyses. The results of statistical analyses of a_c are similar to that of a_0 .

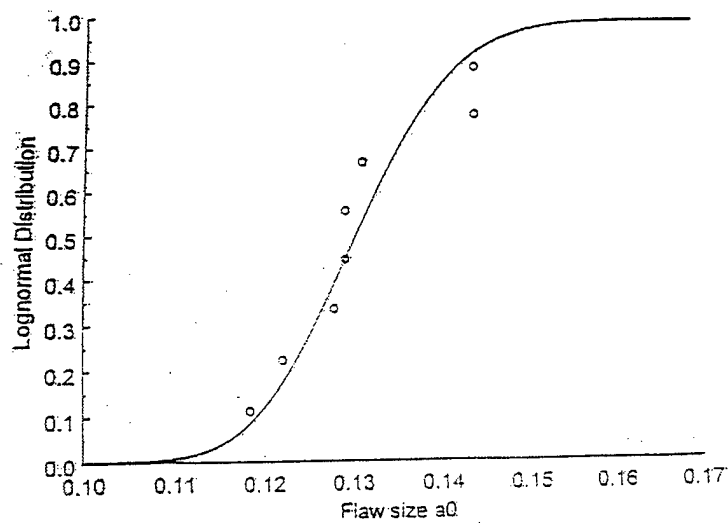


Fig 1: Lognormal Distribution Plot for a_0

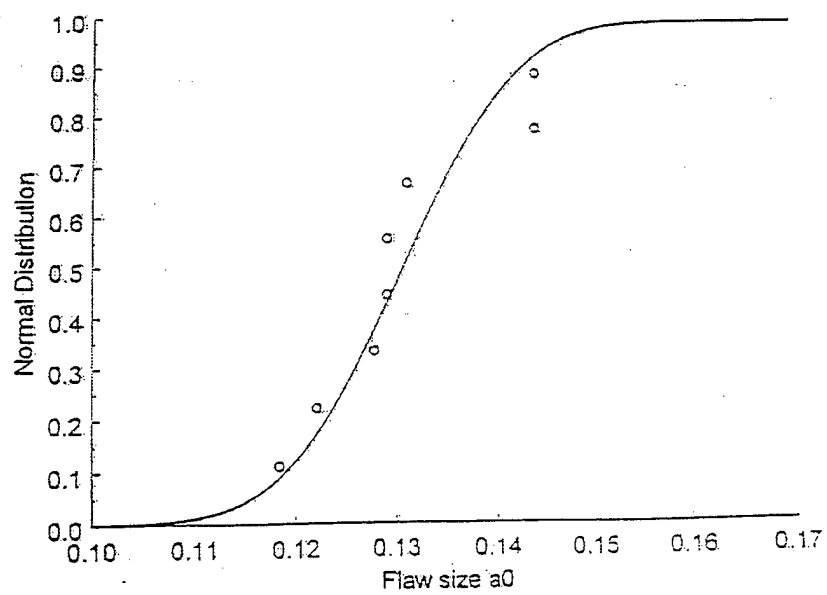


Fig 2: Normal Distribution Plot for a_0

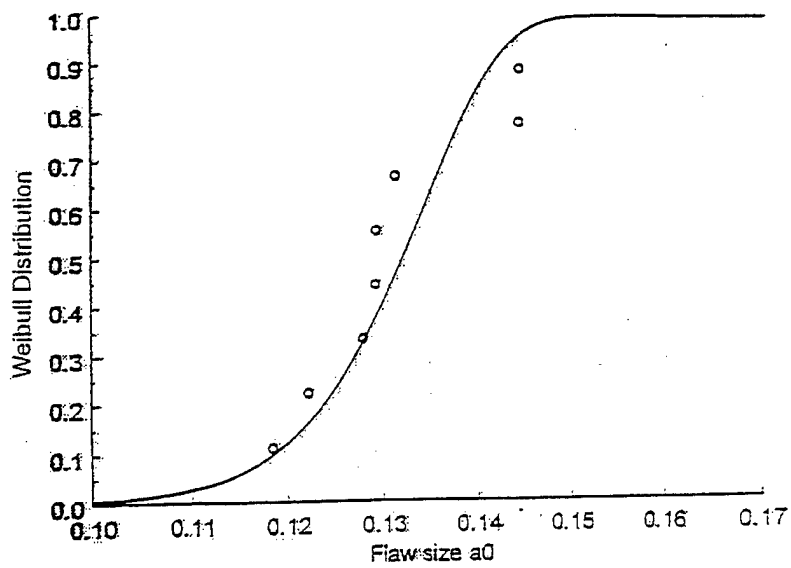


Fig 3: Weibull Distribution Plot for a_0

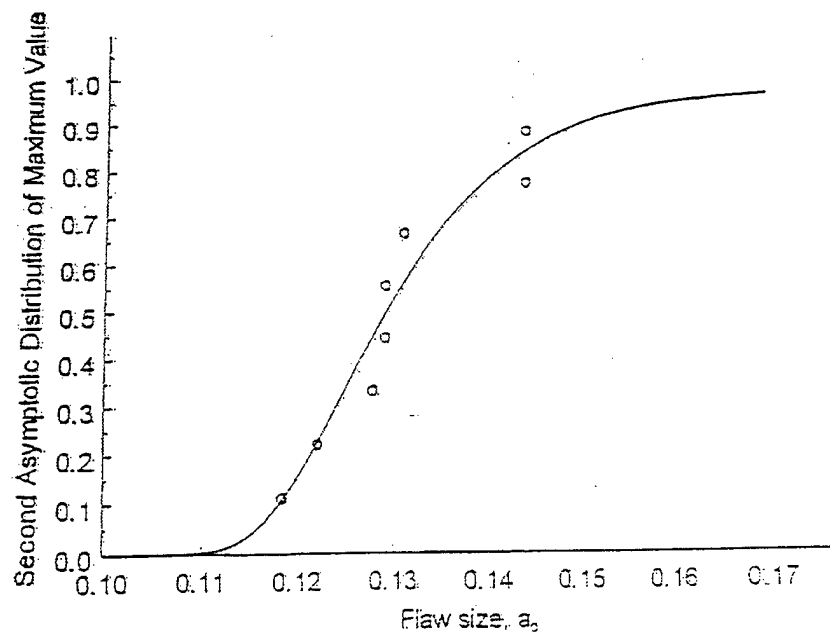


Fig. 4 Second Asymptotic Distribution Plot for a_0

The effect of specimen thickness on the initial flaw size and the critical flaw size were investigated. The results of analyses show that for specimen thicknesses equal to 0.2 in., 0.5 in., and 1.0 in., the variations among a_0 and a_c are small and the average values a_0 and a_c are 0.126 in. and 0.132 in. For specimen thickness equal to 1.5 in. the average values of a_0 and a_c are 0.144 in. and 0.158 in., and they are 14% and 20% higher than that when the specimen thickness is equal to or less than 1.0 in. The increase in a_0 and a_c in thicker specimen is probably due to the size effect of this material. Unlike the brittle metallic materials, this high toughness particulate composite material shows higher strength when the size of the specimen is increased. The size effect of this material is a subject of further research. Considering the highly nonhomogeneous nature of the highly filled particulate composite and for the engineering application purpose, it is reasonable to assume that a_0 and a_c are independent of the specimen thickness. Under this assumption, the average value and the coefficient of variation of a_0 (a_c) are 0.131 in. (0.146 in.) and 0.0703 (0.054), respectively.

Conclusion

In this study, a method is developed to predict the initial crack length and its statistical distribution function of a particulate composite material subjected to a constant strain rate loading condition. The results of analyses indicate that the equivalent initial flaw size and the critical flaw size follow the second asymptotic distribution of maximum value. It also indicates that, on the first approximation and for the engineering application purpose, it can be assumed that the initial crack length is independent of specimen thickness.

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